National Transportation Safety Board Washington, DC 20594

Last 90 Days: 225 Total Make/Model: 19300 Total Instrument Time: 300

Printed on: 08/03/2014 04:04:10 PM

Brief of Accident

Adopted 08/18/2010

DCA09FA065

Instrument Ratings Airplane

File No. 0	07/13/2009	Charleston ,WV	Aircraft Reg No. N387SW		Time (Local): 17:45 EDT	
Engine Make/Model Aircraft Damage Number of Engines Operating Certificate(s) Name of Carrie	 Substantial 2 Flag Carrier/Domestic SOUTHWEST AIRLINES CO Scheduled; Domestic; Passe 		Crew Pass	Fatal 0 0	Serious 0 0	Minor/None 5 126
	: Nashville, TN : Baltimore, MD : Off Airport/Airstrip	Condition of Light: Weather Info Src: Unknown Basic Weather: Lowest Ceiling: Visibility: Wind Dir/Speed: Temperature (°C): Unk/Nr Precip/Obscuration:				
Pilot-in-Command Age: 53		Flight Time (Hours)				
Certificate(s)/Rating(s) Airline Transport; Commercial; Private; Multi-engine Land; Single-engine Land			Total All Aircraft: 22500 Last 90 Days: 225			

*** Note: NTSB investigators either traveled in support of this investigation or conducted a significant amount of investigative work without any travel, and used data obtained from various sources to prepare this aircraft accident report. ***

Flight data recorder data revealed that the airplane took off and climbed for about 25 minutes to an altitude of approximately 35,000 feet, at which point the cabin altitude warning activated, and the captain disengaged the autopilot. Postincident examination of the airplane revealed fatigue cracking of the fuselage skin near the leading edge of the vertical stabilizer adjacent to the rupture. The fatigue cracking penetrated the fuselage skin and created an approximate 18-inch by 12-inch flap in the skin that depressurized the airplane.

The fuselage skin assembly near the leading edge of the vertical stabilizer was manufactured by bonding two full aluminum sheets together, then selectively chemically milling away pockets (bays) of the inner sheet. Continuous fatigue cracks initiated from multiple origins on the inner surface of the skin adjacent to the step formed at the edge of the chemically milled area and propagated outward.

Following the Southwest Airlines (SWA) flight 2294 event, on September 3, 2009, Boeing issued Service Bulletin (SB) 737-53A1301, calling for repetitive external inspections to detect cracks in the fuselage skin along the chemically milled step at stringers S-1 and S-2 right and between BS 827 and BS 847.

Brief of Accident (Continued)

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(The hole from the SWA event was within those boundaries.) If cracks are detected, operators are to contact Boeing for repair instructions. On January 12, 2010, the Federal Aviation Administration issued Airworthiness Directive 2010-01-09, which mandated the inspection requirements in SB 737-53A1301.

Updated at Aug 18 2010 1:52PM

Brief of Accident (Continued)

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OCCURRENCES

Enroute-cruise - Emergency descent initiated Enroute-cruise - Aircraft structural failure

FINDINGS

Aircraft-Aircraft structures-Fuselage-Plates/skins (aux fuselage)-Fatigue/wear/corrosion - C Aircraft-Aircraft structures-Fuselage-Plates/skins (aux fuselage)-Failure - C

Findings Legend: (C) = Cause, (F) = Factor

The National Transportation Safety Board determines the probable cause(s) of this accident as follows: Fuselage skin failure due to preexisting fatigue at a chemically milled step.